

# **DESIGN OF DUAL AXIS FORCE SENSOR FOR AEROSPACE APPLICATION**

Indikadulla Kankanamge Hasitha Prasanna

(08/9306)



University of Moratuwa, Sri Lanka.  
Electronic Theses & Dissertations  
[www.lib.mrt.ac.lk](http://www.lib.mrt.ac.lk)

Degree of Master of Science

Department of Electrical Engineering

University of Moratuwa

Sri Lanka

May 2016

# **DESIGN OF DUAL AXIS FORCE SENSOR FOR AEROSPACE APPLICATION**

Indikadulla Kankanamge Hasitha Prasanna

(08/9306)



University of Moratuwa, Sri Lanka.  
Electronic Theses & Dissertations  
[www.lib.mrt.ac.lk](http://www.lib.mrt.ac.lk)

Thesis submitted in partial fulfilment of the requirements for the degree Master of  
Science in Industrial Automation

Department of Electrical Engineering

University of Moratuwa

Sri Lanka

May 2016

## DECLARATION

I declare that this is my own work and this dissertation does not incorporate without acknowledgement any material previously submitted for a Degree or Diploma in any other University or institute of higher learning and to the best of my knowledge and belief it does not contain any material previously published or written by another person except where the acknowledgement is made in the text.

Also, I hereby grant to University of Moratuwa the non-exclusive right to reproduce and distribute my dissertation, in whole or in part in print, electronic or other medium. I retain the right to use this content in whole or part in future works (such as articles or books).



University of Moratuwa, Sri Lanka.  
Electronic Theses & Dissertations  
[www.lib.mrt.ac.lk](http://www.lib.mrt.ac.lk)

I K H Prasanna

Date:

The above candidate has carried out research for the Masters Dissertation under my supervision.

Prof. Nalin Wickramarachchi

Date:

## Abstract

This work reports on the design procedure of a dual axis force sensor for aerospace applications. System functionality of the force sensor should comply with many reliability aspects peculiar to aerospace industry rather than just sensing the applied force. Final design of the dual axis force sensor is based on three preliminary design concepts and test data of fabricated models. This report discusses descriptively how to come up with new ideas through these models. Mathematical model of the sensor is used to verify design outcomes. Furthermore this work presents the practical circumstances faced during fabricating and testing. Analysis of results are also discussed in the report and comparison of the first three models included in the report.

Functional requirements were fine tuned in the final design compared to the first three design concepts. Major requirement was to reduce the cross sensitivity when it came to the final design. As desired cross sensitivity was 2% of the applied load, Final design enabled to achieve 2.21% pitch cross sensitivity and 3.84% roll cross sensitivity. It was considerable reduction of the cross sensitivity. Non linearity value was reduced by 65.79% and 38.46% pitch and roll respectively. Achieved non linearity value was 0.065% and 0.08% in pitch and roll direction respectively. Hysteresis also reduced by 73.91% in pitch direction and 21.43% in roll direction.

Output of the Wheatstone bridge has to be reduced in order to decrease the cross sensitivity. This required more amplification, causing the reading and the noises to be amplified at the same time. It was required to have more signal conditioning that was a drawback of the system developed



## Acknowledgement

I am heartily thankful to my supervisor, Prof. Nalin Wickramarachchi, whose encouragement, guidance and support from the initial to the final level enabled me to develop an understanding of the subject.

My grateful thanks also go to both Mr. Andy Royal, General Manager, Aero sense technologies (pvt) ltd, United Kingdom, and Mr. Christian Nilsson, Former General Manager, Aero sense technologies (pvt) ltd, Sri Lanka, for giving me permission to conduct internal project as my M.Sc. project, all help me providing all necessary materials and facilities in time and Great ideas and advise throughout the project. This thesis would not have been possible unless my immediate boss Mr. Upul Tennakoon, manager, Technical service center, Aero sense technologies (pvt) ltd, Sri Lanka. He has made available his support in a number of ways. I offer my regards and blessings to all of Aero sense staff who supported me in any aspect during the completion of the project.

Lastly, I should thank many individuals, friends and colleagues who have not been mentioned here personally in making this educational process a success. May be I could not have made it without their supports.



University of Moratuwa, Sri Lanka.  
Electronic Theses & Dissertations  
[www.lib.mrt.ac.lk](http://www.lib.mrt.ac.lk)

## TABLE OF CONTENTS

DECLARATION .....	i
Abstract .....	ii
Acknowledgement .....	iii
TABLE OF CONTENTS .....	iv
LIST OF FIGURES .....	vi
LIST OF TABLES .....	viii
CHAPTER 1 .....	1
1. Introduction .....	1
1.1 Force measurement .....	1
1.2 Mettle foil strain gauge based force sensors .....	2
1.2.1 Mettle foil strain gauges .....	2
1.2.2 Strain gauge base force sensors .....	3
1.3 Flight Controls .....	4
1.3.1 Axes of rotations .....	4
1.3.2 Control surfaces .....	4
1.3.3 Cockpit Control Devices .....	7
1.3.3.1 Primary controls .....	7
1.3.3.2 Secondary controls .....	8
1.3.4 Basic flight control systems .....	9
1.3.4.1 Mechanical flight control systems .....	9
1.3.4.2 Hydro-mechanical .....	10
1.3.4.3 Fly-by-wire control systems .....	10
CHAPTER 2 .....	11
2. Problem Identification .....	11
2.1 Introduction .....	11
2.2 Design specifications .....	11
2.3 Performance testing .....	13
2.3.1 Dead load test .....	13
2.3.2 Parameters of performance .....	14
I. Cross sensitivity percentage .....	14
II. Nonlinearity .....	15
III. Hysteresis .....	15

CHAPTER 3.....	17
3. Preliminary Design Concept .....	17
3.1 Conceptual Designs .....	17
3.1.1 Material Selection.....	17
3.1.2 Three structural Designs .....	17
3.2 Fabrications.....	20
3.3 Practical issues .....	22
3.4 Tests and test results .....	23
3.5 Test results analysis .....	26
3.6 Summery .....	39
CHAPTER 4.....	40
4. Final Design .....	40
4.1 Design concept.....	40
4.2 Mathematical model.....	41
4.3 Outcomes of the mathematical model.....	46
4.4 Fabrications.....	48
4.5 Tests and test results .....	49
CHAPTER 5.....	55
5. Results and Analysis .....	55
5.1 Final results and analysis .....	55
5.2 Conclusion .....	55
5.3 Further development .....	56
References.....	58
Appendix A.....	59
Appendix B.....	60



University of Moratuwa, Sri Lanka.  
 Electronic Theses & Dissertations  
[www.lib.mrt.ac.lk](http://www.lib.mrt.ac.lk)

## LIST OF FIGURES

Figure 1.4: Basic parts of the strain gauge.....	2
Figure 1.5: Example of load cells .....	4
Figure 1.1: Main axis of rotation of the airplane .....	6
Figure 1.2: placement of main control surfaces of the airplane.....	8
Figure 1.3: (a) Yoke controller, (b) Center stick controller, (c) Side stick controller ...	9
Figure 2.1: Dead Load test rig .....	14
Figure 2.2: Real output curves and ideal output curve against load .....	16
Figure 3.1: Concept 1 solid model.....	18
Figure 3.2: Strain gauge placement on single column.....	18
Figure 3.3: Concept 2 solid model.....	19
Figure 3.4: Concept 3 solid model.....	19
Figure 3.5: Strain gauge position marking process.....	20
Figure 3.6: Strain gauge prepared for apply bonding adhesive .....	21
Figure 3.7: Bonding clamp mechanism .....	21
Figure 3.8: Example of sensor wiring.....	22
Figure 3.9: Examples of Damaged strain gauges during fabrications .....	23
Figure 3.10: Percentage cross sensitivity graph of pitch axis, concept 1.....	26
Figure 3.11: Percentage non-linearity graph of pitch axis, concept 1 .....	27
Figure 3.12: Percentage hysteresis graph of pitch axis, concept 1 .....	28
Figure 3.13: Percentage cross sensitivity graph of roll axis, concept 1 .....	28
Figure 3.14: Percentage non linearity graph of roll axis, concept 1 .....	29
Figure 3.15: Percentage hysteresis graph of roll axis, concept 1 .....	30
Figure 3.16: Percentage cross sensitivity graph of pitch axis, concept 2.....	30
Figure 3.17: Percentage non linearity graph of pitch axis, concept 2.....	31
Figure 3.18: Percentage hysteresis values of pitch axis, concept 2 .....	32
Figure 3.19: Percentage cross sensitivity graph of roll axis, concept 2.....	32
Figure 3.20: Percentage non linearity graph of roll axis, concept 2 .....	33
Figure 3.21: Percentage hysteresis graph of roll axis, concept 2.....	34
Figure 3.22: Percentage cross sensitivity graph of pitch axis, concept 3.....	34
Figure 3.23: Percentage non linearity graph of pitch axis, concept 3.....	35
Figure 3.24: Percentage hysteresis graph of pitch axis, concept 3 .....	36
Figure 3.25: Percentage cross sensitivity graph of roll axis, concept 3.....	36
Figure 3.26: Percentage non linearity graph of roll axis, concept 3 .....	37
Figure 3.27: Percentage hysteresis graph of roll axis, concept 3.....	38
Figure 4.1: Solid modal of concept 4, (a) Explode view of the solid model, (b) Integrated solid model.....	40
Figure 4.2: Deformed shape of the clamped guided beam when load applied to in direction perpendicular to the width of column. ....	41
Figure 4.3: Simplified body diagram for mathematical model of outer sensor when pitch load applied .....	42



Figure 4.4: Deformed shape of the clamped guided beam when load applied to in direction parallel to the width of column. ....	43
Figure 4.5: Simplified body diagram for mathematical model of inner sensor when pitch load applied .....	43
Figure 4.6: Simplified body diagram for mathematical model of inner sensor when roll load applied .....	45
Figure 4.7 Simplified body diagram for mathematical model of outer sensor when roll load applied .....	45
Figure 4.8: Strain variation with length of the beam and thickness of the beam.....	47
Figure 4.9: Cross strain variation with length of the beam and thickness of the beam	47
Figure 4.10: Use of bondable terminal in wiring .....	48
Figure 4.11: Percentage cross sensitivity graph of pitch axis, concept 4.....	50
Figure 4.12: Percentage non linearity graph of pitch axis, concept 4.....	51
Figure 4.13: Percentage hysteresis graph of pitch axis, concept 4 .....	51
Figure 4.14: Percentage cross sensitivity graph of roll axis, concept 4.....	52
Figure 4.15: Percentage non linearity graph of roll axis, concept 4 .....	53
Figure 4.16: Percentage hysteresis graph of roll axis, concept 4.....	54
Figure 5.1: Concept 5 design model. (a) Exploded view of sensor concept 5 model without outer casing. (b) Pitch axis load sell in concept 4 design. (c) Roll axis load sell in concept 4 design. (d) Load separator part witch separate effect on sensor parts by roll axis load and pitch axis load. (e) Close view of assembled sensor load separating mechanism.....	57



## LIST OF TABLES

Table 2.1: Applicable load conditions for the DAFSA.....	12
Table 2.2: Cross axis forces sensitivity requirements of DAFSA .....	12
Table 3.1: Load test results of Concept 1, Pitch direction .....	24
Table 3.2: Load test results of concept 1, Roll direction .....	24
Table 3.3: Load test results of Concept 2, Pitch direction .....	24
Table 3.4: Load test results of concept 2, Roll direction .....	25
Table 3.5: Load test results of Concept 3, Pitch direction .....	25
Table 3.6: Load test results of concept 3, Roll direction .....	25
Table 3.7: Percentage cross sensitivity values of pitch axis, concept 1.....	26
Table 3.8: Percentage non linearity values of pitch axis, concept 1 .....	27
Table 3.9: Percentage hysteresis values of pitch axis, concept 1.....	27
Table 3.10: Percentage cross sensitivity values of roll axis, concept 1 .....	28
Table 3.11: Percentage non linearity values of roll axis, concept 1.....	29
Table 3.12: Percentage hysteresis values of roll axis, concept 1 .....	29
Table 3.13: Percentage cross sensitivity values of pitch axis, concept 2.....	30
Table 3.14: Percentage non linearity values of pitch axis, concept 2 .....	31
Table 3.15: Percentage hysteresis values of pitch axis, concept 2.....	31
Table 3.16: Percentage cross sensitivity values of roll axis, concept 2 .....	32
Table 3.17: Percentage non linearity values of roll axis, concept 2.....	33
Table 3.18: Percentage hysteresis values of roll axis, concept 2 .....	33
Table 3.19: Percentage cross sensitivity values of pitch axis, concept 3.....	34
Table 3.20: Percentage non linearity values of pitch axis, concept 3 .....	35
Table 3.21: Percentage hysteresis values of pitch axis, concept 3.....	35
Table 3.22: Percentage cross sensitivity values of roll axis, concept 3 .....	36
Table 3.23: Percentage non linearity values of roll axis, concept 3.....	37
Table 3.24: Percentage hysteresis values of roll axis, concept 3 .....	37
Table 3.25: Summery of analysis results .....	39
Table 4.1: Load test results of concept 4, pitch direction. ....	49
Table 4.2: Load test results of concept 4, roll direction.....	49
Table 4.3: Percentage cross sensitivity values of pitch axis, concept 4.....	50
Table 4.4: Percentage non linearity value of pitch axis, concept 4.....	50
Table 4.5: Percentage hysteresis value of pitch axis, concept 4 .....	51
Table 4.6: Percentage cross sensitivity values of roll axis, concept 4. ....	52
Table 4.7: Percentage non linearity value of roll axis, concept 4. ....	53
Table 4.8: Percentage hysteresis value of roll axis, concept 4.....	53
Table 5.1: Results summary of concept 4 and concept 3.....	55